

# PH 205 SOLUTIONS TO MIDTERM 2

① CENTRAL FORCE  $\vec{F} = -\frac{\alpha}{r^2} e^{-\beta r} \hat{r}$

THIS CAN IN PRINCIPLE BE DERIVED FROM A POTENTIAL  $V(r) = -\int_{\infty}^r \vec{F} \cdot d\vec{r} = \alpha \int_{\infty}^r \frac{e^{-\beta r}}{r^2} dr$

THE KEY IS THAT WE DO NOT ACTUALLY HAVE TO DO THE INTEGRAL!

THE ENERGY, IN POLAR COORDINATES IS  $E = \frac{1}{2} m \dot{r}^2 + \frac{1}{2} m r^2 \dot{\theta}^2 + V(r) = \text{CONST}$

ANGULAR MOMENTUM IS ALSO CONSERVED:  $L = m r^2 \dot{\theta} = \text{CONST}$

$$\therefore E = \underbrace{\frac{1}{2} m \dot{r}^2}_{T_{\text{EFF}}} + \underbrace{\frac{L^2}{2 m r^2}}_{V_{\text{EFF}}} + V(r)$$

THE CONDITION FOR AN EQUILIBRIUM CIRCULAR ORBIT IS  $V'_{\text{EFF}}(r_0) = 0$

THE CONDITION FOR STABILITY OF SMALL OSCILLATIONS IS  $V''_{\text{EFF}}(r_0) > 0$

$$V'_{\text{EFF}} = -\frac{L^2}{m r^3} + \frac{\alpha}{r^2} e^{-\beta r}$$

$$V''_{\text{EFF}} = \frac{3L^2}{m r^4} - \frac{2\alpha}{r^3} e^{-\beta r} - \frac{\alpha\beta}{r^2} e^{-\beta r}$$

$$V'_{\text{EFF}}(r_0) = 0 \Rightarrow \frac{L^2}{m r_0} = \alpha e^{-\beta r_0}$$

THE EQUILIBRIUM ANGULAR VELOCITY  $\Omega$  SATISFIES  $L = m r_0^2 \Omega$

SO  $m \Omega^2 r_0 = \frac{\alpha e^{-\beta r_0}}{r_0^2}$  (WHICH ALSO FOLLOWS FROM  $F=ma$  AT ONCE)

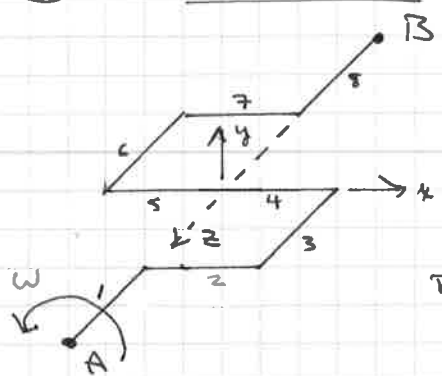
$$V''_{\text{EFF}}(r_0) = 3m\Omega^2 - m\Omega^2 r_0 \left( \frac{2}{r_0} + \beta \right) = m\Omega^2 (1 - r_0 \beta)$$

$$\omega^2_{\text{osc}} = \frac{V''_{\text{EFF}}(r_0)}{m} = \Omega^2 (1 - r_0 \beta) \Rightarrow \omega = \Omega \sqrt{1 - r_0 \beta}$$

CLEARLY FOR STABILITY WE MUST HAVE

$$r_0 < \frac{1}{\beta}$$

## ② THE CRANKSHAFT



WE CAN USE SEVERAL METHODS. BUT SINCE THE CM IS AT REST,  $\vec{F}_A + \vec{F}_B = 0$  IN ALL METHODS

### a) ROTATING FRAME.

THE CRANKSHAFT IS AT REST, BUT WE EXPERIENCE THE CENTRIFUGAL FORCE. BY THE SYMMETRY,  $\sum \vec{F}_{\text{CENTRIFUGAL}} = 0$

BUT THE TORQUE DUE TO THE CENTRIFUGAL FORCE DOES NOT VANISH!

rod #	$\vec{F}_{centric}$	pt of application	$\vec{N} = \text{torque}$
1	0		0
2	$M \frac{a}{2} \omega^2 \hat{x}$	$(\frac{a}{2}, 0, a)$	$M \frac{a^2}{2} \omega^2 \hat{y}$
3	$M a \omega^2 \hat{x}$	$(a, 0, a/2)$	$M a^2/2 \omega^2 \hat{y}$
4	$M \frac{a}{2} \omega^2 \hat{x}$	$(a/2, 0, 0)$	0
5	$-M \frac{a}{2} \omega^2 \hat{x}$	$(-a/2, 0, 0)$	0
6	$-M a \omega^2 \hat{x}$	$(-a, 0, -a/2)$	$M a^2/2 \omega^2 \hat{y}$
7	$-M \frac{a}{2} \omega^2 \hat{x}$	$(-a/2, 0, -a)$	$M a^2/2 \omega^2 \hat{y}$
8	0		0

$\vec{N}_c = 2 M a^2 \omega^2 \hat{y}$

now  $\vec{r}_A \times \vec{F}_A + \vec{r}_B \times \vec{F}_B + \vec{N}_c = 0$

$2 \vec{r}_A \times \vec{F}_A = -2 M a^2 \omega^2 \hat{y}$        $\vec{r}_A = 2a \hat{z}$

$\Rightarrow \boxed{\vec{F}_A = -\vec{F}_B = -\frac{M a \omega^2}{2} \hat{x}}$

b) STAY IN LAB FRAME AND USE  $\vec{N} = \frac{d\vec{L}}{dt}$

where  $\vec{N} = \vec{r}_A \times \vec{F}_A + \vec{r}_B \times \vec{F}_B = 4a \hat{z} \times \vec{F}_A$

RECALL  $\vec{L} = \vec{R} \times \vec{P}_{cm} + \vec{L}_{rel\ to\ cm}$

So we can calculate two pieces of  $\vec{L}$  due to each rod

rod	$\vec{r} \times \vec{p}$	$\vec{L}_{rel}$
1	0	0
2	$(\frac{a}{2}, 0, a) \times (0, M \frac{M a \omega}{2}, 0) = (-\frac{M M a^2 \omega}{2}, 0, \frac{M M a \omega^2}{4})$	$\frac{1}{12} M a^2 \omega \hat{z}$
3	$(a, 0, \frac{a}{2}) \times (0, M M a \omega, 0) = (-M M a^2 \omega/2, 0, M M a \omega^2)$	0
4	$(\frac{a}{2}, 0, 0) \times (0, M M a \omega/2, 0) = (0, 0, M M a^2 \omega/4)$	$\frac{1}{12} M a^2 \omega \hat{z}$
5	$(-a/2, 0, 0) \times (0, -M M a \omega/2, 0) = (0, 0, M M a^2 \omega/4)$	$\frac{1}{12} M a^2 \omega \hat{z}$
6	$(-a, 0, -a/2) \times (0, -M M a \omega, 0) = (-M M a^2 \omega/2, 0, M M a \omega^2)$	0
7	$(-a/2, 0, -a) \times (0, -M M a \omega/2, 0) = (-M M a^2 \omega/2, 0, M M a \omega^2/4)$	$\frac{1}{12} M a^2 \omega \hat{z}$
8	0	0

$\vec{L} = -2 M M a^2 \omega \hat{x} + 10/3 M M a^2 \omega \hat{z}$

$\frac{d\vec{L}}{dt} = -2 M M a^2 \omega \frac{d\hat{x}}{dt}$       and  $\frac{d\hat{x}}{dt} = \vec{\omega} \times \hat{x} = \omega \hat{y}$

so  $\vec{N} = -2 M M \omega^2 a^2 \hat{y} = 4a \hat{z} \times \vec{F}_A \Rightarrow \boxed{\vec{F}_A = -\frac{M a \omega^2}{2} \hat{x} = -\vec{F}_B}$

OR we could use  $\vec{L} = \vec{I} \cdot \vec{\omega} = \vec{I} \cdot \hat{z} \omega = I_{xz} \omega \hat{x} + I_{yz} \omega \hat{y} + I_{zz} \omega \hat{z}$

$I_{yz} = 0$  since the crane is in the x-z plane

$I_{zz} = \text{moment about z axis} = 4 \cdot \frac{1}{3} M a^2 + 2 \cdot M a^2 = \frac{10}{3} M a^2$

$I_{ij} = \int \rho dvol (r^2 \delta_{ij} - r_i r_j)$

$I_{xz} = -\int \rho xz \, dvol$  WE WORK THIS OUT FOR THE MASS ONE BY ONE

$I_{xz}(1) = 0 = I_{xz}(8)$

$I_{xz}(2) = -\frac{M}{a} \cdot a \int_0^a x \, dx = -\frac{1}{2} M a^2$

$I_{xz}(3) = -\frac{M}{a} \cdot a \int_0^a z \, dz = -\frac{1}{2} M a^2$

$I_{xz}(4) = 0 = I_{xz}(5)$

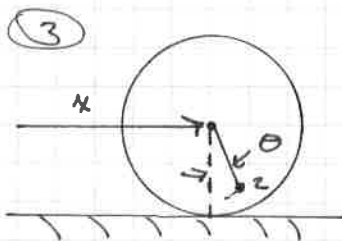
$I_{xz}(6) = -\frac{M}{a} (-a) \int_{-a}^0 z \, dz = -\frac{1}{2} M a^2$

$I_{xz}(7) = -\frac{M}{a} (-a) \int_{-a}^0 x \, dx = -\frac{1}{2} M a^2$

so  $I_{xz} = -2 M a^2$

AND  $\vec{L} = -2 M a^2 \omega \hat{x} + \frac{10}{3} M a^2 \omega \hat{z}$  AS BEFORE.

THERE IS NO NEED TO DIAGONALIZE THE INERTIAL TENSOR.



I CHOOSE  $x$  OF C.M. OF HOOP  
 $\theta$  OF MASS  $2$  TO THE VERTICAL  
 AS COORDS

2 DEGREES OF FREEDOM  $\Rightarrow$  2 MODES

BUT IF WE THINK A BIT, WE NOTE THAT THE HOOP CAN ROLL WITHOUT SLIPPING AT CONSTANT VELOCITY, WHILE  $\theta = 0$  ALWAYS  $\Rightarrow$  NO OSCILLATION.

$\therefore$  WE MAY EXPECT THAT  $\omega = 0$  FOR ONE MODE!

IN THE OTHER MODE WE MIGHT EXPECT  $x$  AND  $\theta$  TO OSCILLATE OPPOSITELY....

$L = T + V$        $T_1 = \frac{1}{2} M_1 \dot{x}^2 + \frac{1}{2} I \omega_{HOOP}^2 = M_1 \dot{x}^2$        $T_2 = \frac{1}{2} M_2 (\dot{x}_2^2 + \dot{y}_2^2)$

$x_2 = x + l \sin \theta$        $\dot{x}_2 = \dot{x} + l \cos \theta \dot{\theta}$

$y_2 = r - l \cos \theta$        $\dot{y}_2 = l \sin \theta \dot{\theta}$

$T_2 = \frac{1}{2} M_2 (\dot{x}^2 + l^2 \dot{\theta}^2 + 2 \dot{x} l \cos \theta \dot{\theta})$

$V = M_2 g y_2 = M_2 g (r - l \cos \theta)$

$L = \frac{1}{2} (2M_1 + M_2) \dot{x}^2 + M_2 \dot{x} l \cos \theta \dot{\theta} + \frac{1}{2} M_2 l^2 \dot{\theta}^2 + M_2 g l \cos \theta$

$\frac{\partial L}{\partial \dot{x}} = (2M_1 + M_2) \dot{x} + M_2 l \cos \theta \dot{\theta} = \text{CONST}$  SINCE  $\frac{\partial L}{\partial x} = 0$

$\frac{\partial L}{\partial \dot{\theta}} = M_2 \dot{x} l \cos \theta + M_2 l^2 \dot{\theta}$

$\frac{d}{dt} \frac{\partial L}{\partial \dot{\theta}} = M_2 \ddot{x} l \cos \theta - M_2 \dot{x} l \sin \theta \dot{\theta} + M_2 l^2 \ddot{\theta} = \frac{\partial L}{\partial \theta} = -M_2 \dot{x} l \sin \theta \dot{\theta} - M_2 g l \sin \theta$

# PH 205 MIDTERM 2

FROM THE X EQUATION

$$\ddot{x} = \frac{-m_2}{2m_1+m_2} l \cos \theta \ddot{\theta} + \frac{m_2}{2m_1+m_2} l \sin \theta \dot{\theta}^2$$

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$$s.o. \ddot{\theta} \left( 1 - \frac{m_2}{2m_1+m_2} \cos^2 \theta \right) + \frac{m_2}{2m_1+m_2} \sin \theta \cos \theta \dot{\theta}^2 = - \frac{g}{l} \sin \theta$$

THIS CAN CERTAINLY BE SATISFIED BY  $\theta = 0 \Rightarrow \dot{x} = \text{CONST} ; x = vt$

IF  $\theta$  NOT CONSTANT, WE MAKE THE SMALL ANGLE APPROXIMATION

$$\frac{2m_1}{2m_1+m_2} \ddot{\theta} \approx - \frac{g}{l} \theta$$

$$\Rightarrow \theta = A \cos \omega t \quad \text{WHERE}$$

$$\omega = \sqrt{\frac{2m_1+m_2}{2m_1} \frac{g}{l}}$$

$$\dot{x} \approx - \frac{m_2}{2m_1+m_2} l \dot{\theta} \Rightarrow x = - \frac{m_2}{2m_1+m_2} l \theta = - \frac{m_2}{2m_1+m_2} l A \cos \omega t$$

$$x_{\text{MAX}} = + \frac{m_2}{2m_1+m_2} l \theta_{\text{MAX}}$$

$x$  AND  $\theta$  ARE IN PHASE  $180^\circ$  OUT OF PHASE AS EXPECTED

NOTE THAT IF THE C.M. OF THE SYSTEM DID NOT MOVE HORIZONTALLY, THEN WE EXPECT

$$m_1 x + m_2 (x + l \theta) = 0$$

$$\text{OR } x = - \frac{m_2}{m_1+m_2} l \theta$$

BUT THE C.M. IS NOT NECESSARILY AT REST BECAUSE OF THE CONSTRAINT FORCE!